# EVALUATION OF TRIBOLOGICAL PROPERTIES OF HIGH-STRENGTH MATERIAL WITH DEPOSITED CrN LAYER

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### Abstract

Two basic requirements such as strength of materials and/or material requirements for the lowest mass, must be meet in the development trends of new high-strength material with high tensile strength, with fracture toughness, with ductility, with better energy-efficient, with high security and cheaper production. The highest growth is recorded in TRIP steels and alloys of Maraging - currently among the strongest materials. Availability of the martensitic steels HCM 18 (EU: 1.2709, DIN: X3NiCoMoTi18-9-5, US: 18 Maraging 300) and its features: high strength, hardness, easy workability, nearly zero dimensional changes after heat treatment, designate these steels for their using in highly stressed components in the aerospace, automotive industries, etc. The aim of the article is microstructure evaluation of HCM 18, the roughness of the substrate and selected CrN coating, evaluation of tribological properties (scratch test) and the wear rate (volume loss of material).

Keywords: high-strength material, HCM 18, thin layer, CrN coating, tribological properties.

### 1 Introduction

Weight reduction of machinery, improving the dynamic parameters in varying loading conditions in conjunction with their energy savings in operation can be efficiently carried by integrating multiple high-tech features in one product. Increasing the durability and reliability of machine components requires finding innovative ways of creating advanced materials with high yield strength, tensile strength, fracture toughness and ductility. Therefore, the development of new high strength steels and their other improvements continues. Particularly in the automotive, aviation, aerospace and shipbuilding industry will required lighter and stronger structural components, with improved energy efficiency while reducing emissions and pollution, increased security and cheaper production.

The primary importance of the use of high-strength material is to reduce the supporting cross-sections (satisfy the requirement of sufficient rigidity), and the total weight reduction of machinery. There are two basic criteria (not enough to know the absolute values of Rm and Rp0.2 – Table 1) [1, 2]:

a) high-strength materials – it takes into account the stiffness of the body:

$$\frac{R_m; R_p 0, 2}{E} \ge \frac{1}{150}$$
(1)

b) material requirements for the lowest weight (aircraft, etc.) – demanding cyclically loaded components, significantly it manifests loading of centrifugal forces and inertia forces (ρ - density):

$$\frac{R_m; R_p 0, 2}{\rho} \ge 0, 2 \tag{2}$$

| Materials    | R <sub>m</sub> [MPa] | <b>R</b> <sub>p</sub> 0,2 [MPa] | R <sub>p</sub> 0,2/E [-] | R <sub>p</sub> 0,2/ρ [Nm/kg] |
|--------------|----------------------|---------------------------------|--------------------------|------------------------------|
| STN 41 6341  | 1800                 | 1450                            | 1/145                    | 0,19                         |
| 98Ni300grade | 1990                 | 1960                            | 1/95                     | 0,24                         |
| TiAl5CrFeMo  | 1500                 | 1400                            | 1/58                     | 0,30                         |

Table 1 A comparison of selected high-strength materials properties according to relationship (1), (2)

Table 1 shows that steel 16341 is beyond the capabilities of conventional steel processing, and therefore the advanced high-strength steels are used in other ways to influence the structure [3, 4, 5]:

- a) TMP thermomechanical processing (purposefully controlled by a combination of shaping and phase transformations), low/high thermomechanical processing LTMP/HTMP, Isoforming,
- b) hardening martensitic matrix (maraging steels: VACO 180 Bolzano, VACUMAR),
- c) deformation induced martensitic transformation (TRIP steels),
- d) same nonferrous alloys [1, 2]:
  - corrosion resisting: on basis Ni Monel, Inconel, Incoloy; on basis Co Stellit, MP, Vitallium,
  - heat resisting: on basis Ni a Co Nimonic, LVN, MAR M, MA.

Group of researchers from Wayne State University's College of Engineering uses computers in the search for advanced materials, where their efforts led to the development of new high-strength bainitic steel with an extremely fine microstructure of ferrite and carbon stabilized austenite by isothermal hardening. This material has double the size of the yield strength, high tensile strength, almost three times the fracture toughness of existing steels, higher fatigue strength and impact resistance, better durability, lower weight. Independent tests have shown excellent results in improvised explosive ballistic attacks, making finds application in military industry [6].

As the available solution can be used tool martensitic steels HCM 18 (19 901) for the purposes of tribological properties evaluation of deposited CrN layer – chemical composition is shown in Table 2.

Table 2 The chemical composition of the steel HCM 18 [Wt. %]

| Steel              | С     | Mn   | Si   | Cr   | Al   | Ni    | Cu | Mo   | Со   | Ti   | Р     | S      |
|--------------------|-------|------|------|------|------|-------|----|------|------|------|-------|--------|
| HCM 18<br>(19 901) | 0,025 | 0,04 | 0,08 | 0,06 | 0,12 | 17,65 | -  | 4,76 | 7,00 | 0,61 | 0,084 | 0,0101 |

Label classification steel (HCM 18), according to US - ,,18 Maraging 300<sup>o</sup>, by EU -  $,,1.2709^{\circ}$ , by DIN -  $,,X3NiCoMoTi18-9-5^{\circ}$ . This steel has a very high strength and surface hardness, is easily workable, the curable up to 54 HRC, has virtually no change in dimensions during heat treatment. The material is commonly used for highly stressed parts in the aerospace industry, pressure vessels made from it gears, extrusion tools, cold, plastic molds and tools for pressure casting, etc. [7, 8, 9].

The aim of the article is microstructure evaluation of HCM 18, the roughness of the substrate and selected CrN coating, evaluation of tribological properties (scratch test) and the wear rate (volume loss of material).

### 2 Materials and methods

Steel test samples of HCM 18 are made by cutting out a circular blank of diameter D = 30 mm (Fig. 1), the chemical composition is shown in Table 2. The samples were quenched in a vacuum at 860 °C and tempered at 500 °C, after heat treatment, the samples were polished with the diamond paste with a particle size 1µm. The parameters of deposition CrN coating were realized in a company LISS, Inc., process parameters are shown in Table 3.



Fig. 1 Steel test samples of HCM 18

| Parameters                 | Units   | CrN coating                |  |  |
|----------------------------|---|----------------------------|--|--|
| Equipment                  | [-]   | Platit 80                  |  |  |
| Process swagger            | [-]   | LARC                       |  |  |
| The voltage at the cathode | [V]   | 50                         |  |  |
| Substrate temperature      | [°C]  | 220                        |  |  |
| Length deposition          | [s]   | 2400                       |  |  |
| Pressure                   | [Pa]  | 4                          |  |  |
| Characteristic cleaning    | [negative polarization of the substrate / deposition / protective atmosphere] | 750V / 200V / Ar55<br>sccm |  |  |

Before evaluating the microstructure of the coating and the substrate it was necessary to prepare the sample, i.e. to polish the sample in the transverse direction of the thin layer metallographic, then the sample was etched (to make the microstructure visibly), the thin layer region and the substrate was collected by means of REM, the thickness of the layer

Table 3 The parameters of deposition CrN coating

was measured and using EDS spectrometer were detected the elements present in the transfer layer of the microregionsubstrate.

Scratch test (indentation test) was performed on UMT Tribolab, the normal force was growing (Fn = 1 to 100 N) at a constant speed as well as movement of the samples was constant. A standard Rockwell diamond indenter tip with a radius of curvature of 0.2 mm was used. An acoustic emission (AE) from the normal force Fn and the coefficient of friction value were recorded during the measurement. Violation of surface morphology was documented by light microscopy and the analysis of these results brought a value of critical load thin layer (adhesive layer).

The tribological properties of thin layer were evaluated on high-temperature tribometer with method of pin-on-disc, the counterpart was a small ball with Al2O3 and with a diameter 6 mm. Tests were performed without lubrication at room temperature and a relative humidity of 33%. The test load was 1 N, speed of the sample was 8 cm/s, and the total distance was 100 m. The calculation of the wear size was made by "ASTM Standard G99-95a" [10], volume loss of the sample surface was determined according to equation (3):

$$V_{u} = 2\pi R \left[ r^{2} \cdot \sin^{-1} \left( \frac{d}{2r} \right) - \left( \frac{d}{4} \right) \cdot \left( 4r^{2} - d^{2} \right)^{\frac{1}{2}} \right]$$
(3)

#### **3** Results and diskussion

The roughness of the sample was evaluated after the chemical - heat treatment, after polishing with diamond paste grain size of 1 micron, i.e. before and after application of the coating on the profilometer Mitutoyo SJ-two hundred and first. An important parameter is the arithmetic mean deviation of surface roughness (Ra = 0,42 microns) and the maximum surface unevenness (Rz = 4,27 mm). These values were measured after the coating application.

The evaluation of the microstructure (Fig. 2), of the elements contained in the thin layer (FIG. 3) and the layer thickness measurement (Table 4) were carried by highly distinctive Scanning Electron Microscopy JEOL JSM7600F.



Fig. 2 Microstructure-substrate interface layer



Fig. 3 Elements contained in the deposited layer CrN

Table 4 The layer thickness on the substrate 18 HCM (19 901)

| HCM 18<br>(19 901) | Units | CrN coating |
|--------------------|-------|-------------|
|                    |       | 1,29        |
| Layer thickness    | [µm]  |             |

The image evaluation scratch tracks highlighting the violation (Fig. 4) and a comparison of the signals of interest during the EA with the behaviour coefficient of friction, was determine a critical loads damaging of the thin layer. The first significant change signal usually indicates the start of the infringement of the coating and the significant change of the friction coefficient during the measurement usually occurs on the first substrate detection., The AE signal was not recorded in the scratch test (Fig. 5), the behaviour and value of the friction coefficient is shown in Fig. 6.





Fig. 5 The behaviour of AE in scratch test

Fig. 6 The behaviour of friction coefficient

During the assessing of the CrN coating on the substrate HCM 18 were the critical load as follows: normal load for the first violation of layers  $F_{n1} = 27$  N, load at layer peels  $F_{n2} = 35$  N.

A functionality of the friction coefficient was measured at a load of 2 and 5 N, the results are presented in Fig. 7 - it shows the differences are minimal, and at a load of 5 N, there is the stabilization of the measurement course to the end of almost constant. The size of wear layers CrN is shown in Fig. 8.



Fig. 7 The behaviour of friction coefficient on HCM 18 subtrate



The evaluation of the tribological properties of CrN deposited on a substrate layer 18 HCM showed impaired adhesion, which may be caused by a very small layer thickness (1.29 mm), the signal from the AE was not recorded. Coefficient of friction layer at a load of 5 N after stabilization showed almost constant value.

# 4 Conclusion

To verify and confirm presented results, find the cause of faults in the measurement of the AE scratch test, it is appropriate to repeat the experiment measurement in order to formulate conclusions with the correct information value.

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